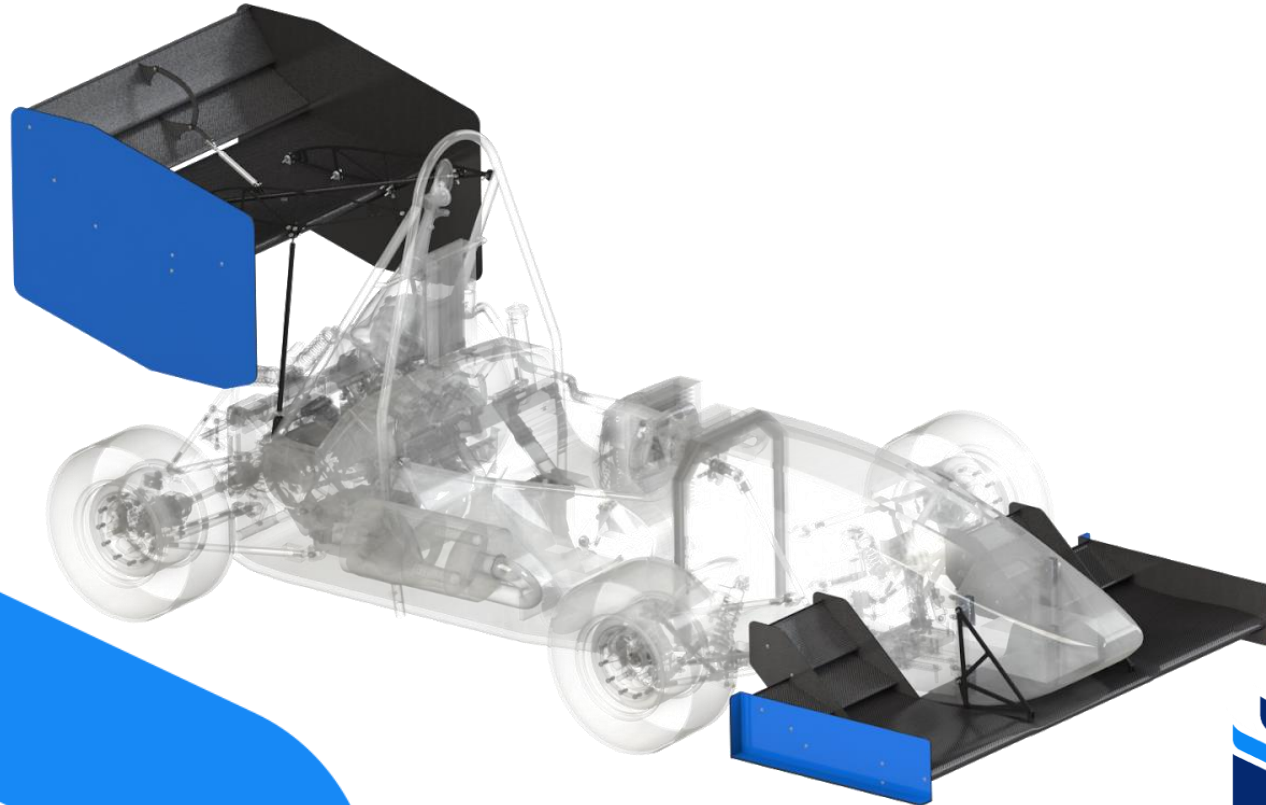


PBR20

Aerodynamics Design Presentation



UMSAE
Polar Bear Racing



**University
of Manitoba**

Section at a Glance

- **$C_l \cdot A = -2.25$**
Front Wing -> -1.21
Rear Wing -> -1.10
(DRS Open -> -0.44)
- **$C_d \cdot A = 1.23$**
Front Wing -> 0.24
Rear Wing -> 0.41
(DRS Open -> 0.08)
- **$L/D = -1.82$**
Front Wing -> -5.00
Rear Wing -> -2.65
(DRS Open -> -5.38)
- **Mass = 10.2 [kg]**
Front Wing -> 4.5 [kg]
Rear Wing -> 5.7 [kg]
- **Frontal Area = 1.05 [m²]**
Front Wing -> +0.05 [m²] (mostly from muffler side)
Rear Wing -> +0.24 [m²]
- **Aerobalance = 50.0%**
Adjustable from 41.5% to 57.0%

Section Goals



- The goal: Increase points at competition

Event	Points Delta
Endurance	+40.4
Fuel	-11.1
Autocross	+25.3
Skidpad	+9.9
Acceleration	+4.6
Cost	-2.2
Total	+66.9

Parameter	Points Delta
Cl*A	+98.2
Cd*A	-18.1
Mass	-4.8

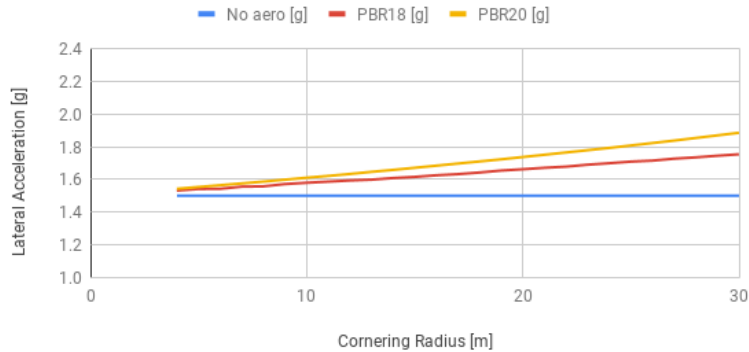
Sensitivities

+0.1 Cl*A gives:	3.58 points	+0.028 Cl*A gives:	1 point
-0.1 Cd*A gives:	2.90 points	-0.034 Cd*A gives:	1 point
-0.1 [kg] gives:	0.04 points	-2.50 [kg] gives:	1 point

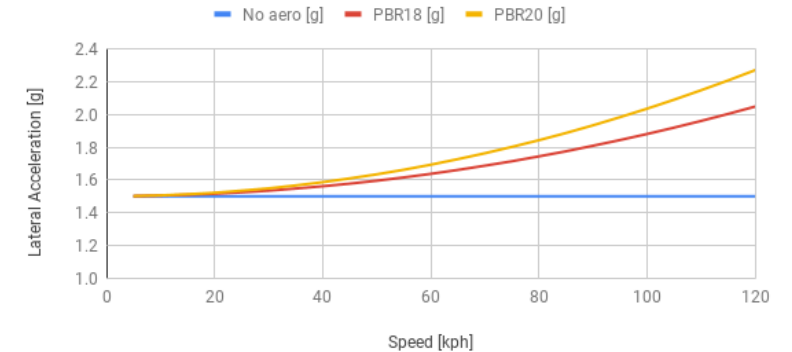
Critical L/D 0.81

Aerodynamic Performance Gains

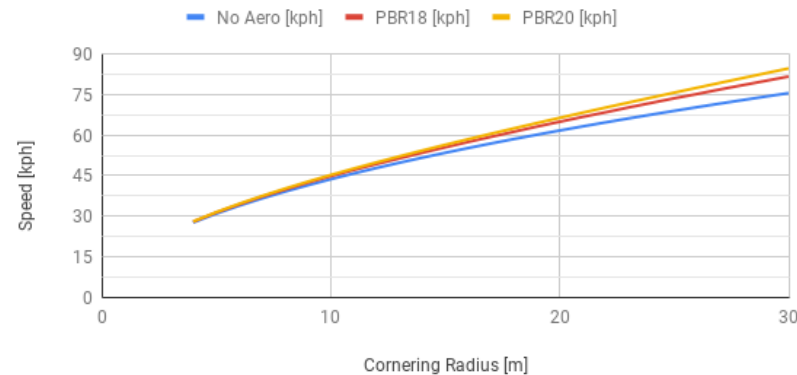
Lateral Acceleration vs Cornering Radius



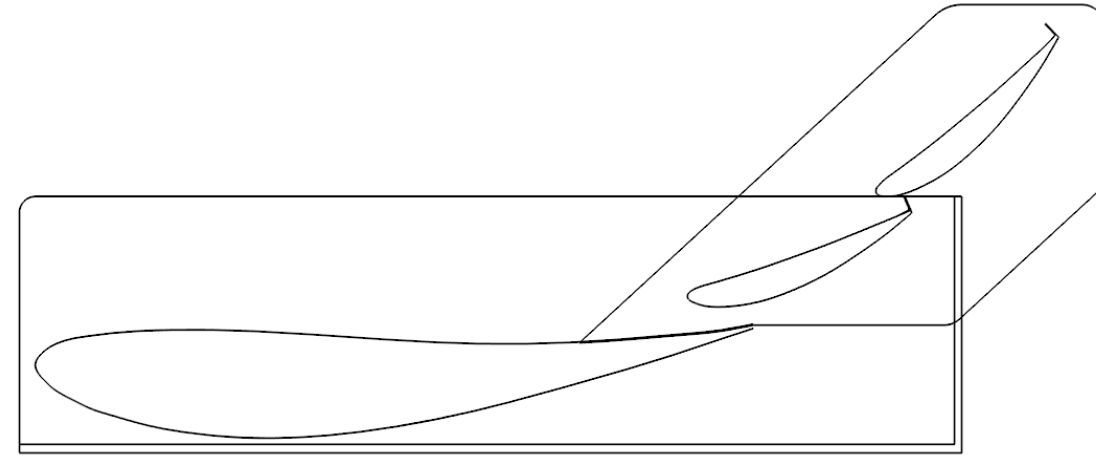
Lateral Acceleration vs Speed



Speed vs Cornering Radius

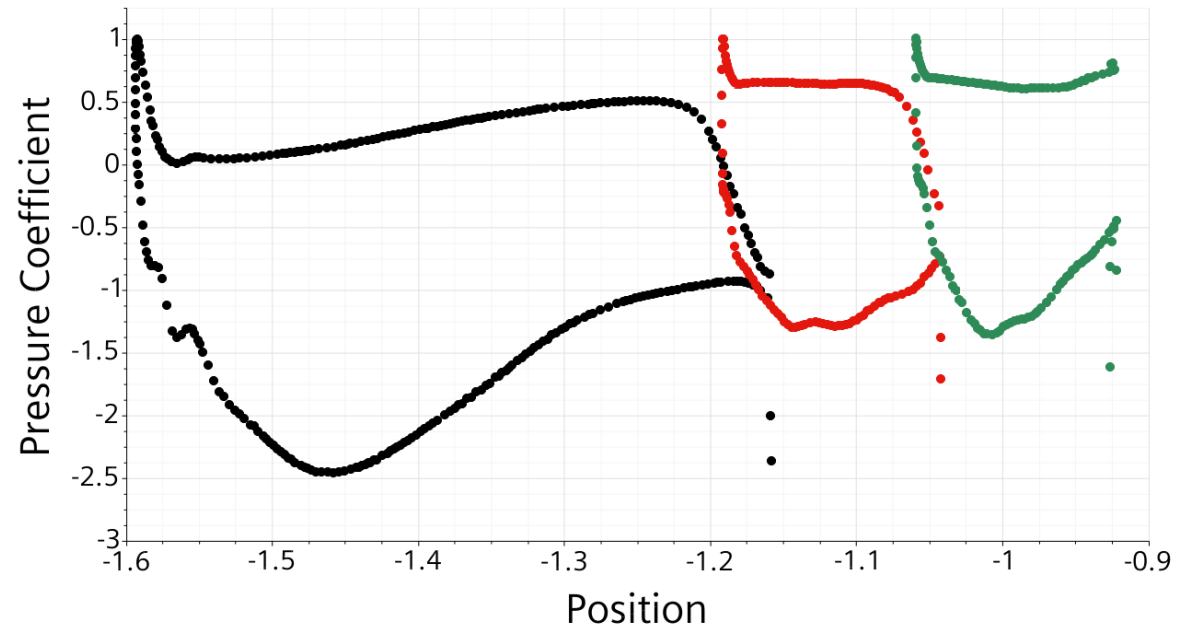


Front Wing

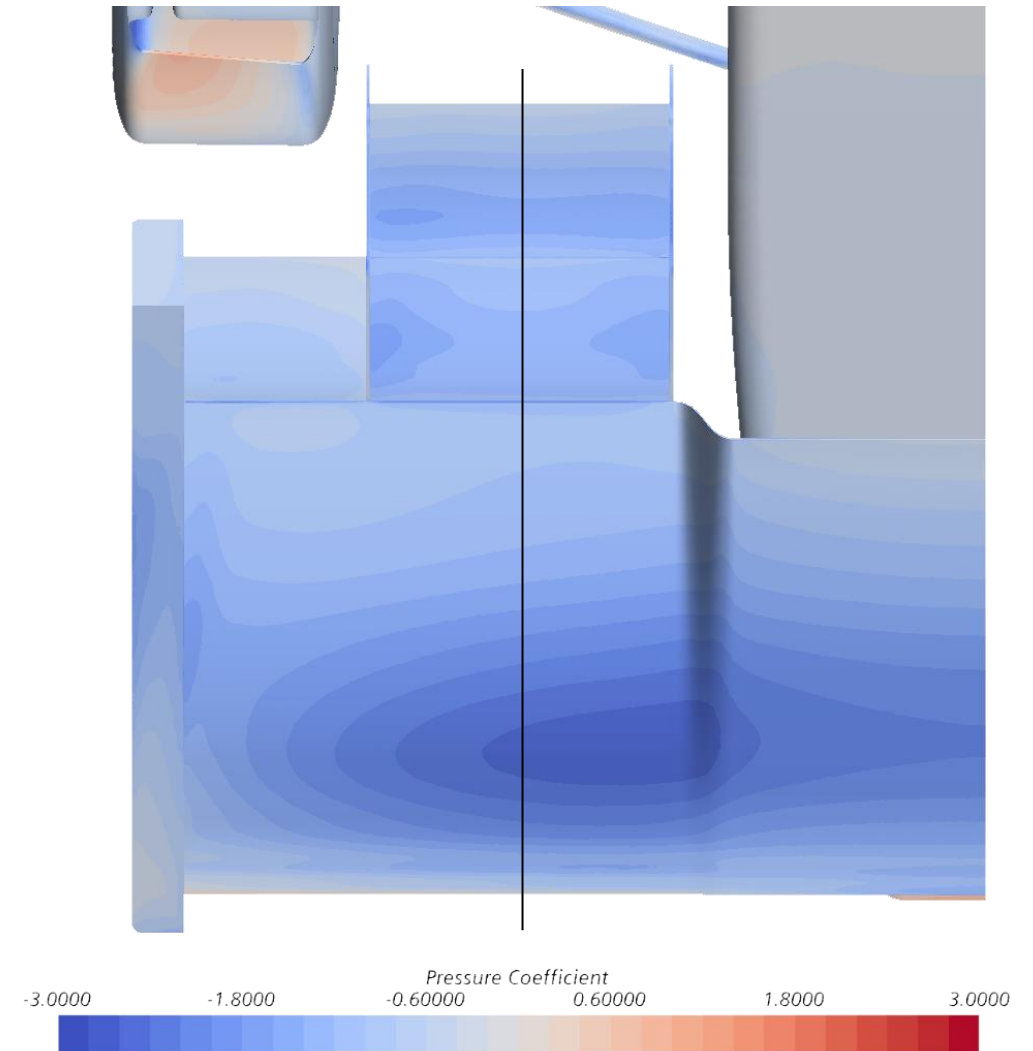


	Cord [mm]	Angle of Attack (relative) [deg]	Slot Gap (ground clearance) [mm]	Slot Overlap [mm]
Center Mainfoil	450	-5	70	n/a
Side Mainfoil	450	3	80	n/a
Flap 1	150	21 (18)	10	15
Flap 2	150	41 (20)	9	15

Front Wing



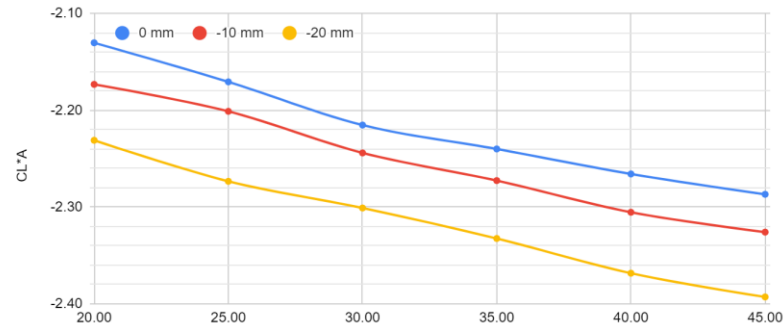
Font Wing Pressure Coefficient Profile, Y380



Front Wing – It moves!

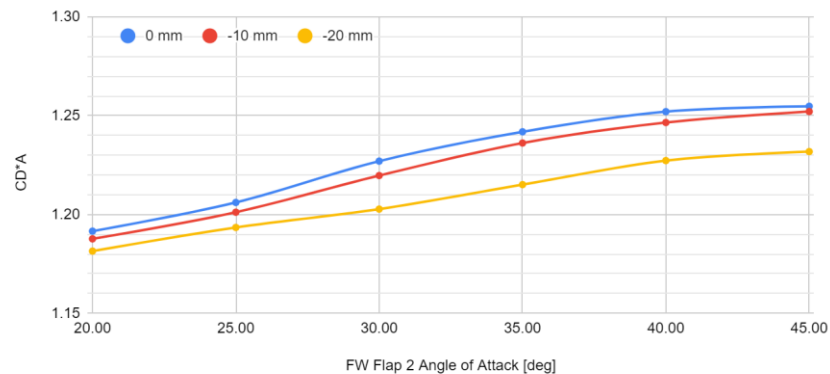
Front Wing Settings, CL*A

Sets = FW Ride Height Adjustment



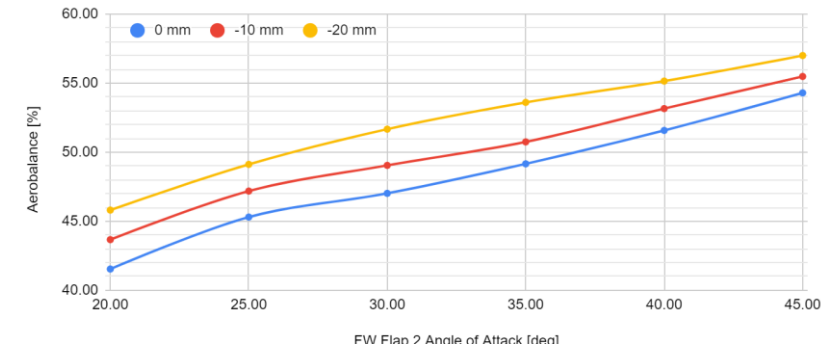
Front Wing Settings, CD*A

Sets = FW Ride Height Adjustment



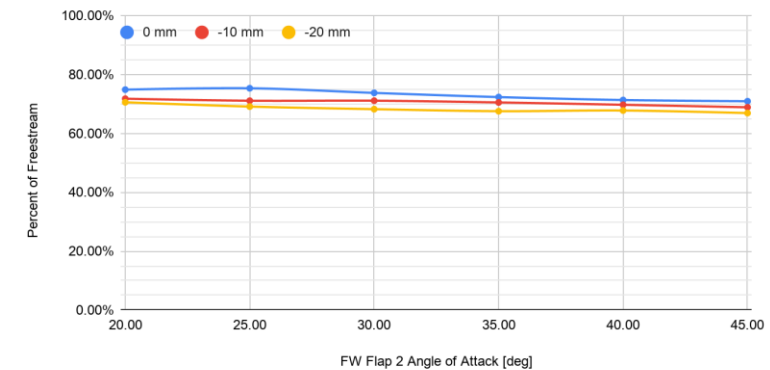
Front Wing Settings, Aerobalance

Sets = FW Ride Height Adjustment

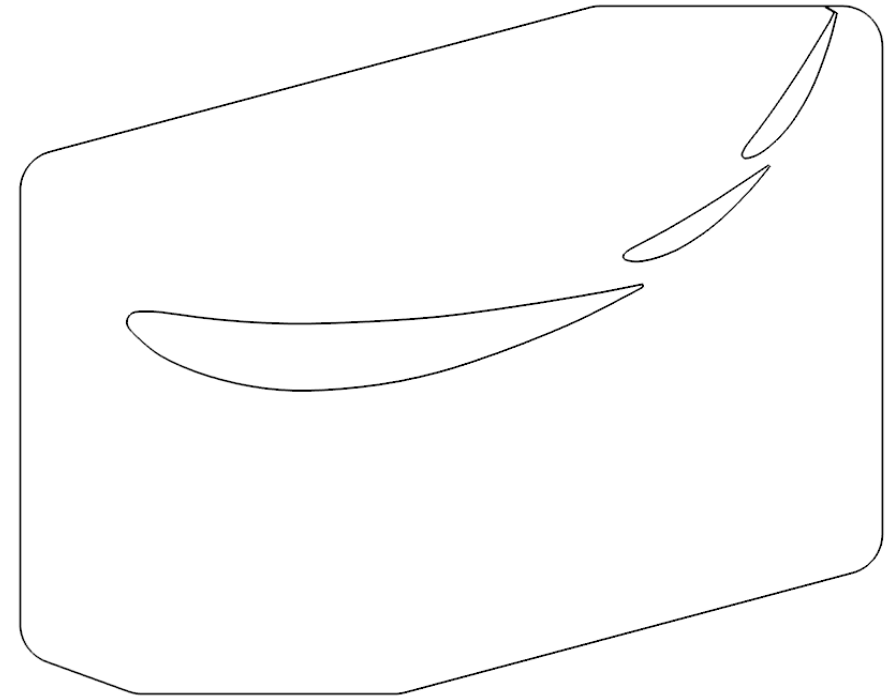


Front Wing Settings, Radiator Velocity (15 m/s)

Sets = FW Ride Height Adjustment

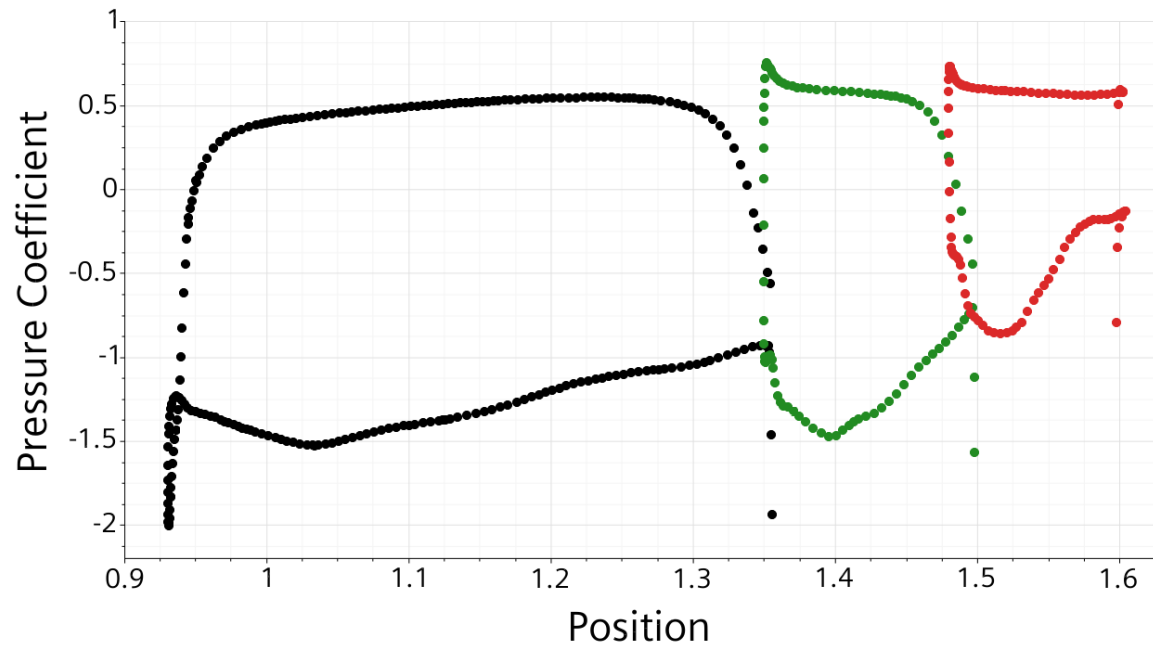


Rear Wing



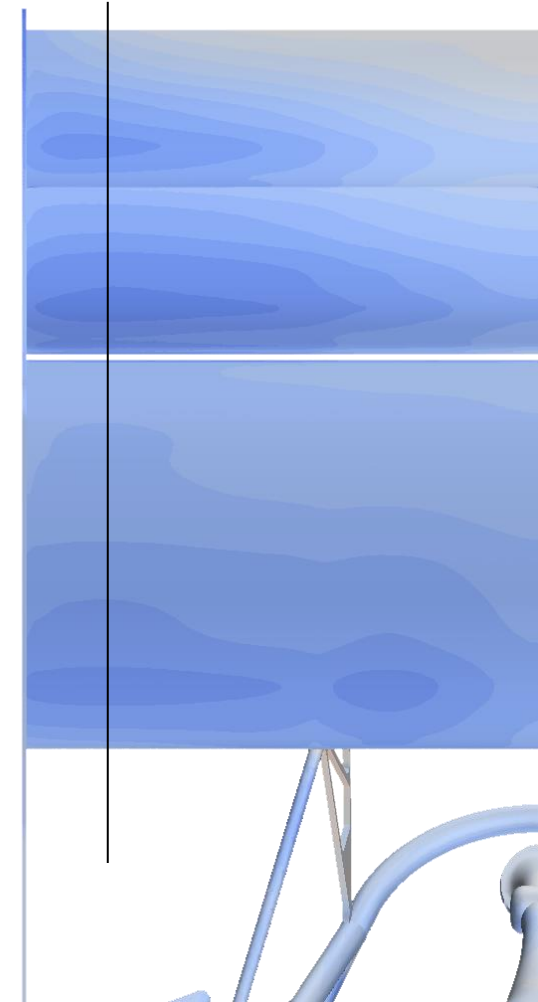
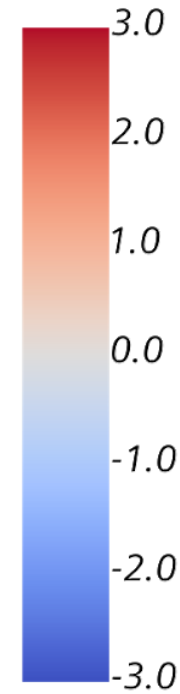
	Cord [mm]	Angle of Attack (relative) [deg]	Slot Gap (ground clearance) [mm]	Slot Overlap [mm]
Mainfoil	450	4	n/a	n/a
Flap 1	150	32.5 (28.5)	18.5	15
Flap 2	150	57.5 (25)	12.5	15

Rear Wing

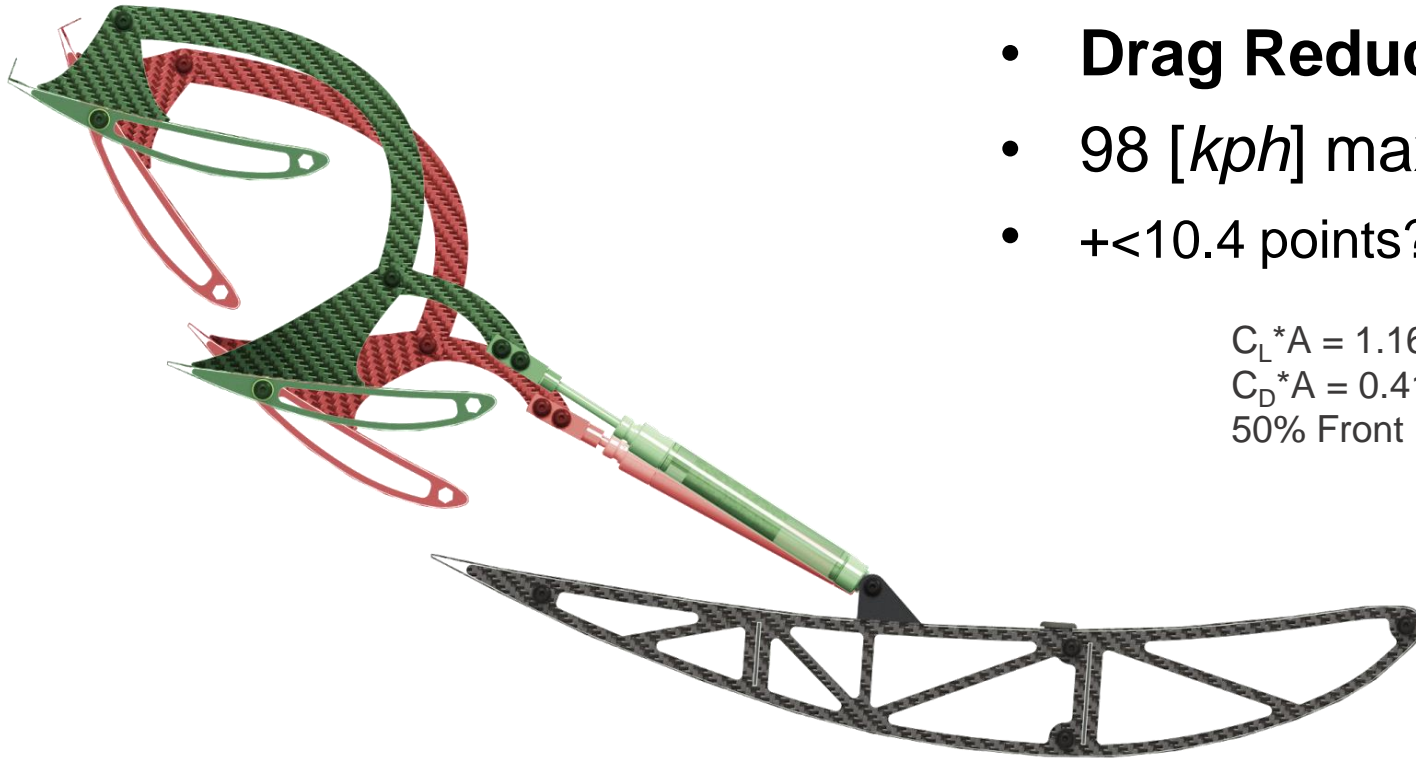


Rear Wing Pressure Coefficient Profile, Y380

Pressure Coefficient



Rear Wing – It also moves!



- **Drag Reduction System**
- 98 [kph] maximum opening speed
- +<10.4 points?

$C_L * A = 1.16$
 $C_D * A = 0.41$
50% Front



$C_L * A = 0.46$ (-0.7, -60%)
 $C_D * A = 0.058$ (-0.352, -80%)
87% Front (+37% balance)

DRS Optimization - % changes in

		Flap 1 Opening [deg]									
		0	-5	-10	-15	-20	-25	-30	-35	-40	
Flap 2 Opening [deg]	0	100	104	108	110	112	111	111	107	101	
	-5	104	108	114	115	116	115	113	110	104	
	-10	108	113	121	121	121	121	118	114	105	
	-15	113	119	124	128	126	127	123	117	110	
	-20	118	123	132	136	136	132	127	122	113	
	-25	125	130	136	142	148	143	136	125	115	
	-30	131	137	143	150	157	158	151	132	119	
	-35	139	146	151	159	166	169	166	147	125	
	-40	147	153	161	167	175	179	175	158	135	
	-45	154	161	169	176	183	192	191	172	147	
	-50	152	160	171	182	191	203	198	180	160	
	-55	144	155	166	178	184	196	196	183	160	
	-60	138	146	155	163	175	178	169	154	137	
	-65	131	137	145	153	156	163	158	138	123	

L/D

		Flap 1 Opening [deg]									
		0	-5	-10	-15	-20	-25	-30	-35	-40	
Flap 2 Opening [deg]	0	41	45	49	52	56	59	62	65	67	
	-5	42	45	50	53	57	59	63	66	68	
	-10	44	47	51	54	58	61	64	67	69	
	-15	45	48	51	56	59	63	65	68	71	
	-20	48	49	51	56	61	64	67	70	72	
	-25	50	51	53	57	61	65	69	71	75	
	-30	53	54	56	58	61	66	70	74	77	
	-35	56	57	58	60	63	67	72	76	80	
	-40	60	61	62	64	66	68	72	76	79	
	-45	67	68	67	68	69	73	75	77	79	
	-50	70	73	75	75	75	77	79	80	81	
	-55	70	73	76	81	83	85	86	86	85	
	-60	70	73	77	79	84	88	92	95	95	
	-65	71	73	76	80	83	88	93	96	97	

Aerobalance

		Flap 1 Opening [deg]									
		0	-5	-10	-15	-20	-25	-30	-35	-40	
Flap 2 Opening [deg]	0	100	88	76	68	61	56	52	49	48	
	-5	93	82	70	63	57	53	49	47	46	
	-10	87	77	66	58	53	49	46	44	43	
	-15	80	72	63	54	50	45	43	41	40	
	-20	73	67	59	51	45	42	40	38	37	
	-25	66	61	55	48	42	37	36	35	34	
	-30	59	54	50	45	39	34	31	31	31	
	-35	52	48	44	40	35	31	28	27	28	
	-40	44	42	38	35	31	28	26	25	26	
	-45	36	33	32	29	27	24	22	22	24	
	-50	33	28	26	23	22	20	19	19	22	
	-55	35	30	25	21	18	16	15	16	18	
	-60	36	31	27	23	19	16	14	14	15	
	-65	37	33	29	24	21	17	15	14	16	

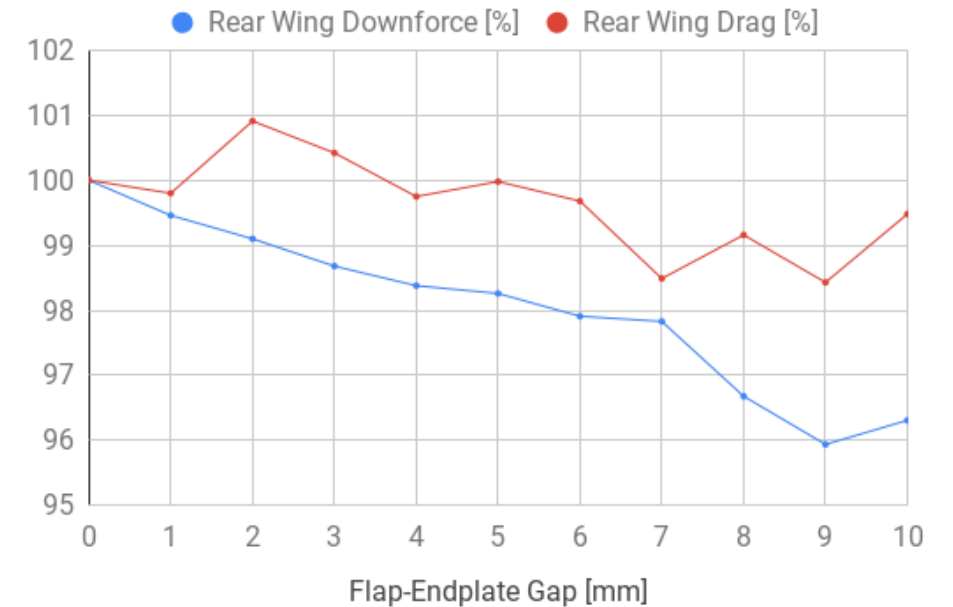
Drag

		Flap 1 Opening [deg]									
		0	-5	-10	-15	-20	-25	-30	-35	-40	
Flap 2 Opening [deg]	0	100	91	82	75	68	62	57	53	48	
	-5	96	89	80	72	66	61	56	51	47	
	-10	94	87	80	71	65	60	54	50	45	
	-15	90	85	79	69	63	58	53	48	44	
	-20	87	83	78	70	61	55	50	46	42	
	-25	83	80	75	69	62	53	48	43	39	
	-30	77	75	71	67	61	54	47	41	37	
	-35	72	70	67	63	59	53	46	40	35	
	-40	65	64	61	58	54	50	45	39	36	
	-45	55	53	54	52	49	46	42	38	36	
	-50	50	45	44	42	42	40	37	35	35	
	-55	50	46	42	37	33	31	30	29	29	
	-60	49	45	41	37	33	28	23	21	21	
	-65	49	45	41	37	33	29	24	20	19	

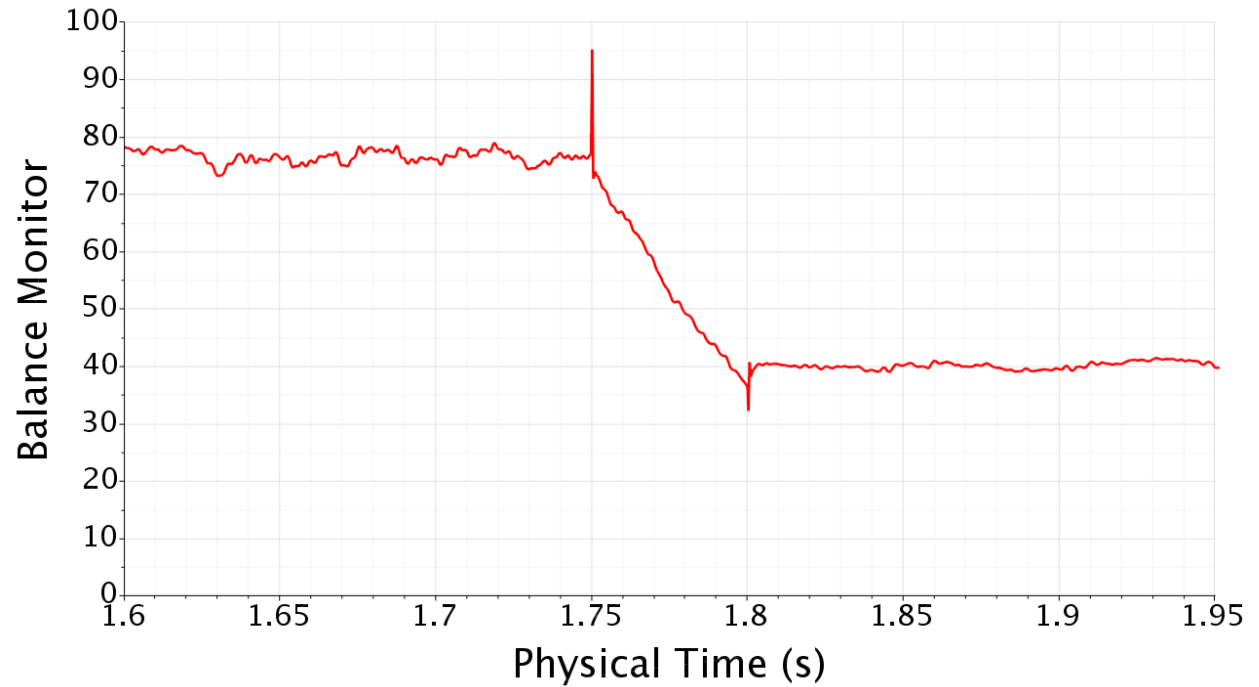
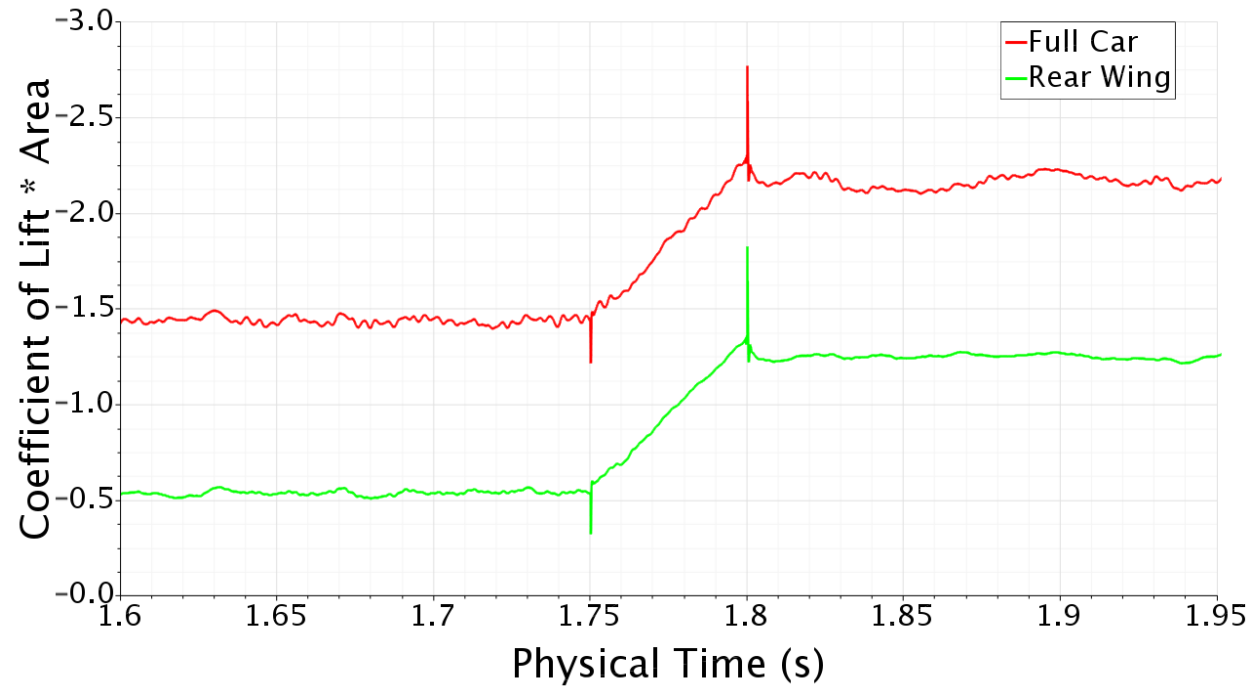
Downforce

DRS Losses

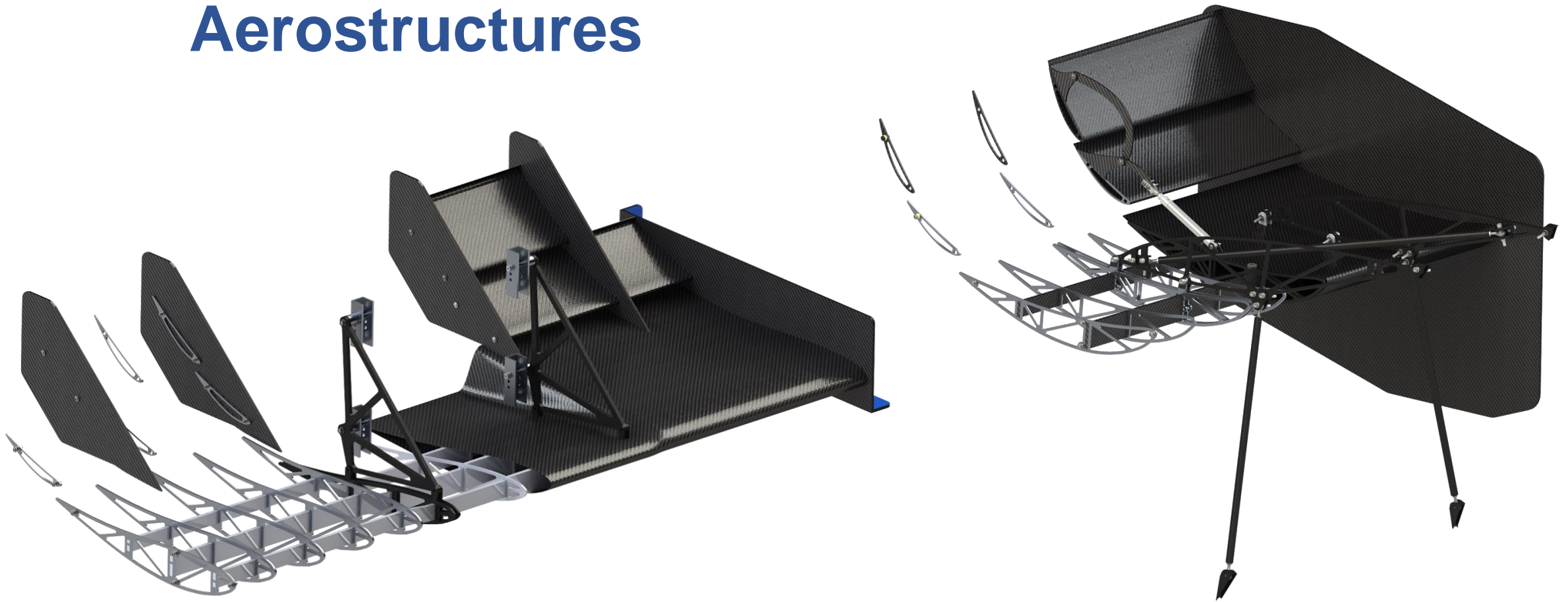
	C_L*A		C_D*A	
	No Linkage	Linkage	No Linkage	Linkage
Mainfoil	0.7687	0.7644 (-0.56%)	0.0699	0.0695 (-0.57%)
Flap 1	0.2739	0.2715 (-0.88%)	0.1604	0.1591 (-0.81%)
Flap 2	0.115	0.1134 (-1.39%)	0.1782	0.1755 (-1.52%)
DRS Linkage	n/a	0.0001	n/a	0.0006
Overall	1.1575	1.1495 (-0.69%)	0.4084	0.4046 (-0.93%)



DRS - Closing



Aerostructures



FEA Inputs

- 35 [*m/s*] top speed (126 [*kph*]) + 15 [*m/s*] wind+gust (54 [*kph*]) = 50 [*m/s*] (180 [*kph*])
- 1.5 Factor of Safety
- Full xyz aero load vector
- Acceleration loading from mass
- Buckling

FEA Load Cases

- **Cornering (max front wing)**
- **Braking (max front wing)**
- **90 deg spin -> rear wing sideforce of 1000 [N] !**
- **180 deg spin -> inconsequential**
- **Front wing jacking / big road bump**
- **Foot on front wing “Claude Case”**
- **Future -> Cone strike**

Aerodynamic Mapping – Ride Height

CL*A		Height Front [mm]				
		10	20	30	40	50
Height Rear [mm]	10	-2.37	-2.25	-2.14	-2.04	-1.93
	20	-2.44	-2.31	-2.21	-2.10	-2.00
	30	-2.51	-2.38	-2.26	-2.16	-2.07
	40	-2.61	-2.45	-2.34	-2.22	-2.12
	50	-2.65	-2.51	-2.39	-2.28	-2.17

Aerobalance Front		Height Front [mm]				
		10	20	30	40	50
Height Rear [mm]	10	56.35	52.15	49.02	45.36	41.47
	20	58.31	54.38	50.74	47.20	43.73
	30	59.67	56.42	52.25	49.03	45.77
	40	59.21	58.24	53.63	50.63	47.39
	50	60.92	59.25	55.45	52.30	49.04

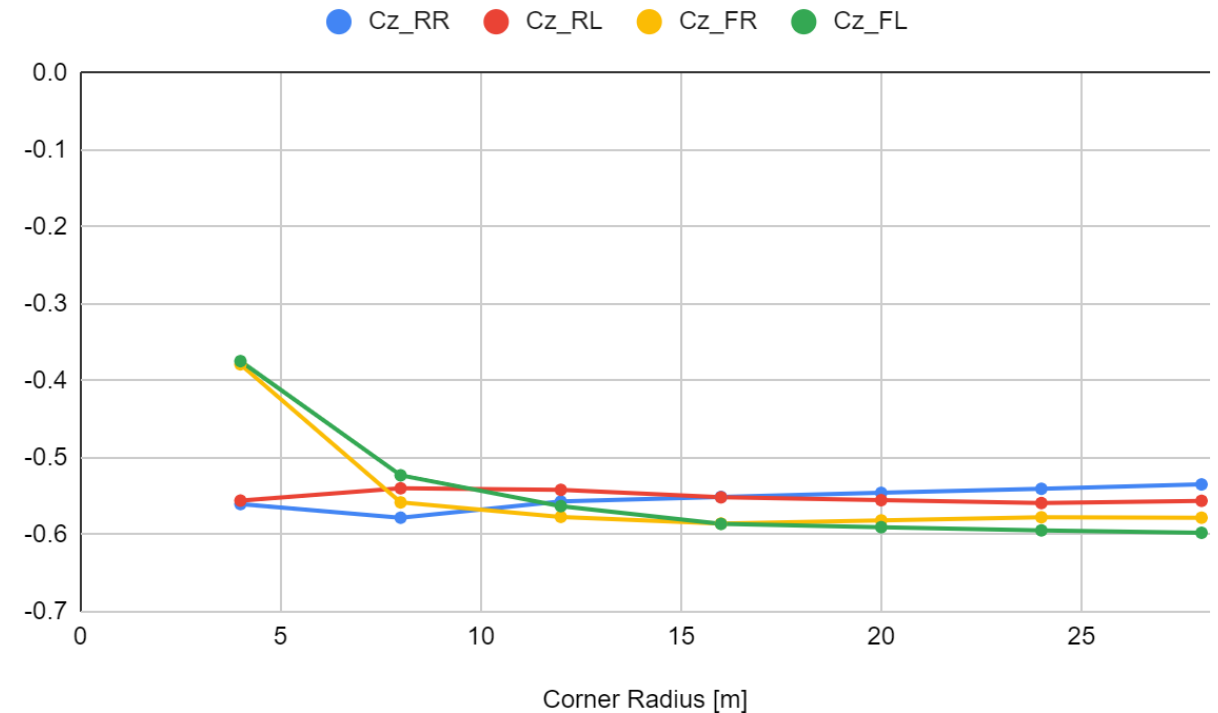
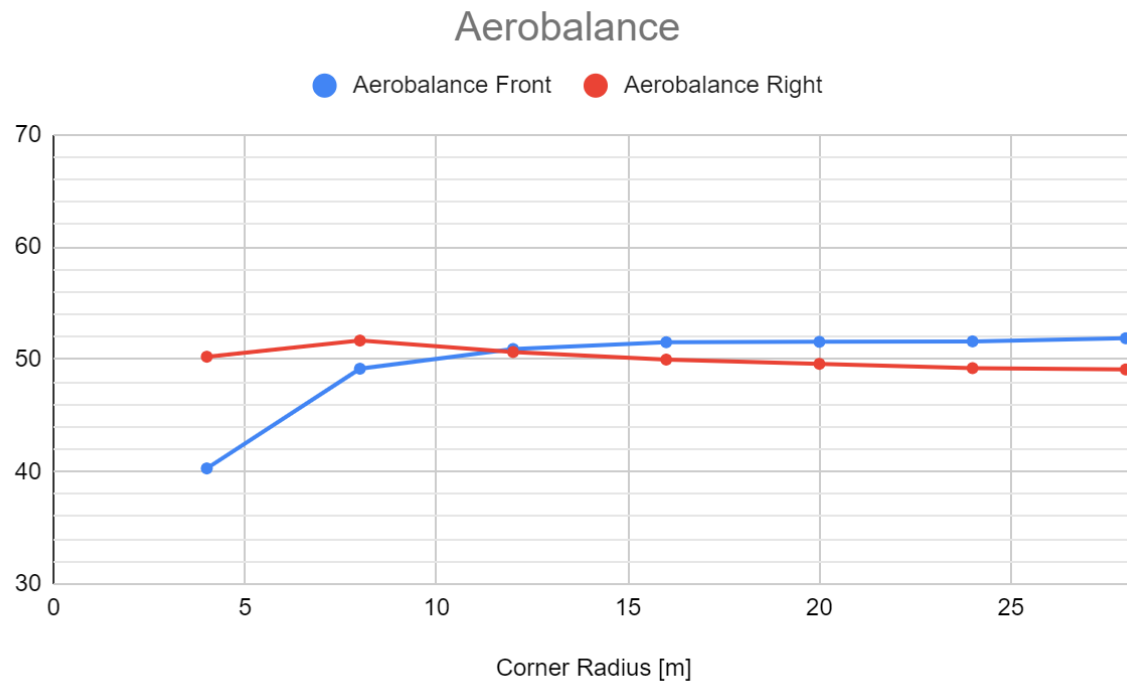
CzF		Height Front [mm]				
		10	20	30	40	50
Height Rear [mm]	10	-1.34	-1.17	-1.05	-0.92	-0.80
	20	-1.42	-1.26	-1.12	-0.99	-0.87
	30	-1.50	-1.34	-1.18	-1.06	-0.95
	40	-1.55	-1.42	-1.26	-1.12	-1.01
	50	-1.61	-1.49	-1.32	-1.19	-1.07

CD*A		Height Front [mm]				
		10	20	30	40	50
Height Rear [mm]	10	1.21	1.22	1.22	1.21	1.20
	20	1.21	1.23	1.23	1.22	1.21
	30	1.21	1.23	1.24	1.23	1.22
	40	1.25	1.22	1.25	1.24	1.23
	50	1.24	1.24	1.25	1.25	1.24

L/D		Height Front [mm]				
		10	20	30	40	50
Height Rear [mm]	10	-1.97	-1.84	-1.76	-1.69	-1.61
	20	-2.02	-1.88	-1.80	-1.73	-1.65
	30	-2.07	-1.93	-1.83	-1.76	-1.69
	40	-2.10	-2.00	-1.88	-1.79	-1.72
	50	-2.14	-2.03	-1.90	-1.82	-1.75

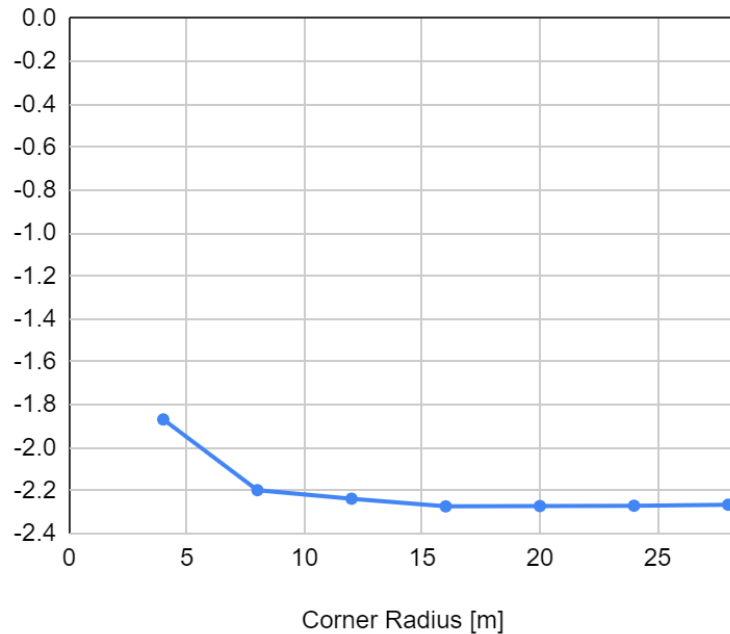
CzR		Height Front [mm]				
		10	20	30	40	50
Height Rear [mm]	10	-1.04	-1.08	-1.09	-1.11	-1.13
	20	-1.02	-1.05	-1.09	-1.11	-1.13
	30	-1.01	-1.04	-1.08	-1.10	-1.12
	40	-1.07	-1.02	-1.09	-1.10	-1.12
	50	-1.04	-1.02	-1.06	-1.09	-1.11

Aerodynamic Mapping - Cornering

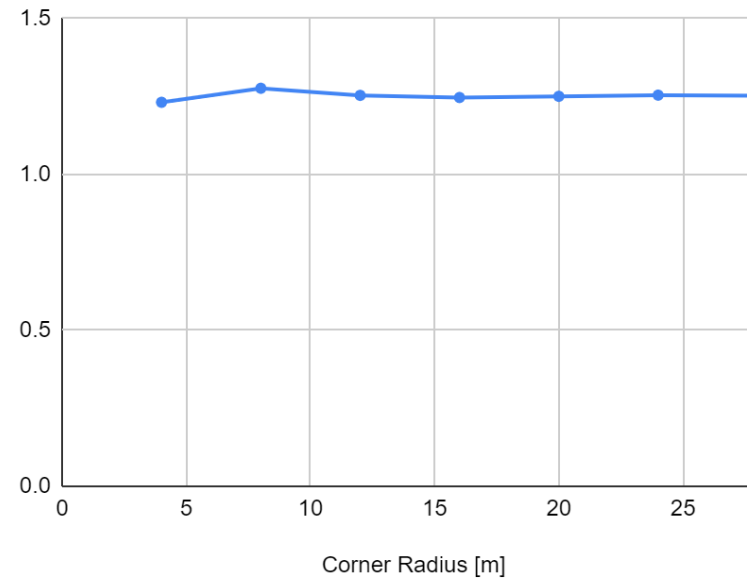


Aerodynamic Mapping - Cornering

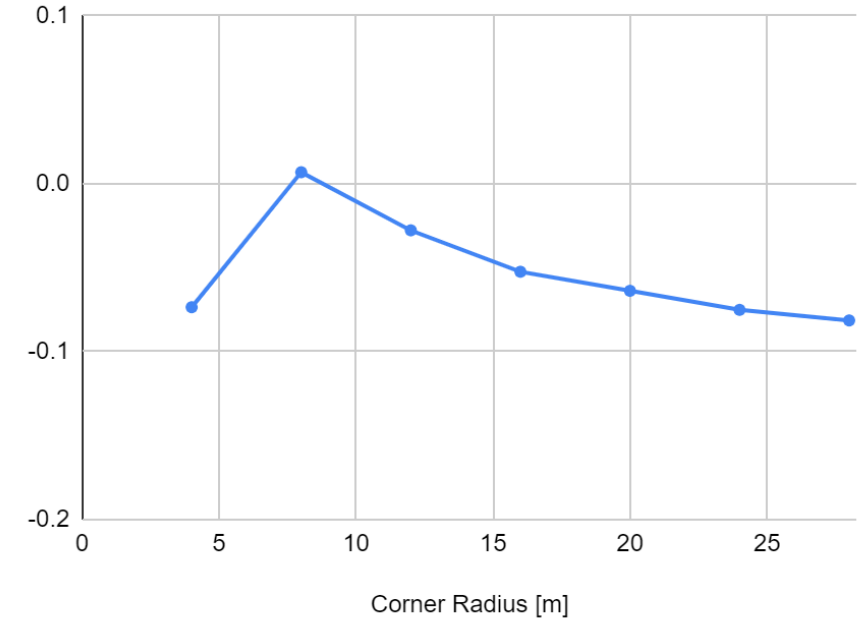
Downforce Coefficient



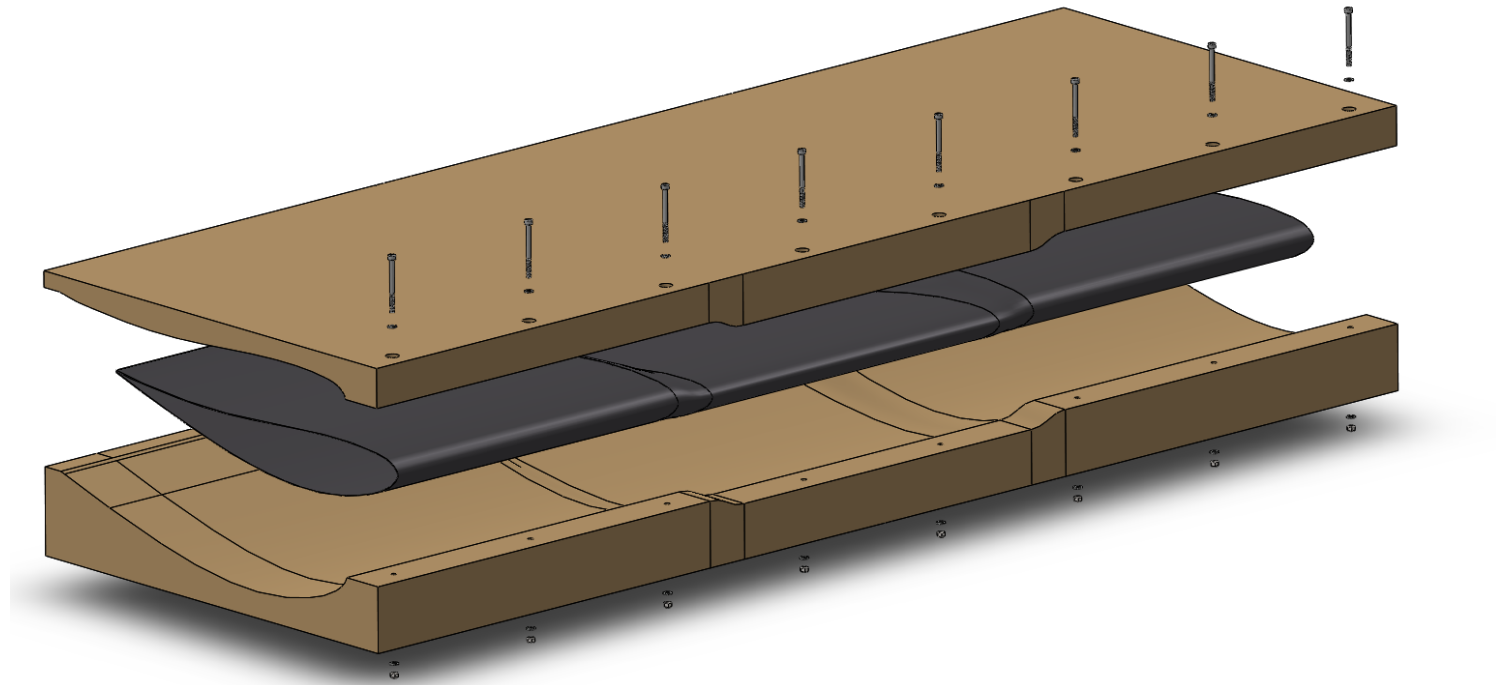
Drag Coefficient



Sideforce Coefficient



Manufacturing



Validation - delayed due to pandemic

- Yarn tests
- Coast-downs
- Suspension potentiometers
- Flow visualization paint